PAGE: 1 PRINT DATE: 01/06/98

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- NON-CIL HARDWARE NUMBER: 05-6Q-2203A -X

SUBSYSTEM NAME: EPD&C - DISPLAYS & CONTROLS

		REVISION: 2	01/06/98
PART DATA			
	PART NAME	PART NUMBER VENDOR NUMBER	
	VENDOR NAME		
LRIJ	: PANEL F7A5 (PRE-MEDS)	V070-730261	
LRJ	: PANEL F6A3 (MEDS)	ME452-0102-7352	
SRU	: SWITCH, TOGGLE	ME452-0102-7352	
\$RŲ	: SWITCH, TOGGLE	ME452-0102-7852	
SRU	: SWITCH, TOGGLE	ME452-0102-8352	

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:

SWITCH, TOGGLE, 3 POLE/2 POSITION - FLIGHT CONTROLLER POWER, COMMANDER'S STATION.

REFERENCE DESIGNATORS:

34V73A7A5S2 (PRE-MEDS)

34V73A6A3S2 (MEDS CONFIGURATION)

QUANTITY OF LIKE ITEMS: 1

ONE ON COMMANDER STATION PNL.

FUNCTION:

ENABLES CREW TO TURN COMMANDER STATION DDU FLIGHT CONTROL POWER SUPPLIES A, B AND C "ON" OR "OFF" FOR RHC, THC, SBTC, RPTA, AND BFC.

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FAILURE MODES EFFECTS ANALYSIS FMEA -- NON-CIL FAILURE MODE

NUMBER: 05-6Q-2203A-01

REVISION#: 2

01/06/98

SUBSYSTEM NAME: EPD&C - DISPLAYS & CONTROLS

LRU: PANEL F7A5 (PRE-MEDS), F6A3 (MEDS)

CRITICALITY OF THIS

ITEM NAME: SWITCH, TOGGLE

FAILURE MODE: 1R3

FAILURE MODE:

FAILS CLOSED IN "OFF" POSITION, SHORT TO CASE AT "ON" POSITION CONTACTS, POLE TO POLE SHORT.

MISSION PHASE:

PL PRE-LAUNCH

LO LIFT-OFF

OO ON-ORBIT

DO DE-ORBIT

LS LANDING/SAFING

VEHICLE/PAYLOAD/KIT EFFECTIVITY:

102 COLUMBIA

103 DISCOVERY

104 ATLANTIS

105 ENDEAVOUR

CAUSE:

CONTAMINATION, INTERNAL STRUCTURAL FAILURE, VIBRATION, SHOCK.

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN

A) PASS

B) PASS

C) PASS

PASS/FAIL RATIONALE:

A)

B)

C)

CORRECTING ACTION: MANUAL

CORRECTING ACTION DESCRIPTION:

AFTER TOTAL (EITHER TWO OR THREE CHANNEL FAILURES) FAILURE OF THE COMMANDERS'S FLIGHT CONTROLLER POWER SWITCH, THE FLIGHT CONTROL FUNCTION SHALL BE TRANSFERRED TO THE PILOT'S STATION.

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FAILURE MODES EFFECTS ANALYSIS (FMEA) -- NON-CIL FAILURE MODE NUMBER: 05-6Q-2203A- 01

- FAILURE EFFECTS -

(A) SUBSYSTEM:

LOSS OF ABILITY TO TURN ON AFFECTED FLIGHT CONTROL POWER SUPPLIES A, B, OR C.

(B) INTERFACING SUBSYSTEM(S):

LOSS OF POWER TO ONE CHANNEL ON EACH OF THE FOLLOWING: RHC, THC, SBTC, RPTA, AND BFC AT THE COMMANDER'S STATION.

(C) MISSION:

FIRST FAILURE - NO EFFECT.

(D) CREW, VEHICLE, AND ELEMENT(S):

POSSIBLE LOSS OF CREW/VEHICLE DUE TO LOSS OF ELECTRICAL POWER TO FLIGHT CONTROLLERS DURING CRITICAL FLIGHT PHASES.

(E) FUNCTIONAL CRITICALITY EFFECTS:

SUCCESS PATHS AFTER FIRST FAILURE - FLIGHT CONTROL CAN BE TRANSFERRED TO PILOT'S STATION. FAILURE OF THE PILOT'S STATION FLIGHT CONTROLLER POWER SWITCH WILL RESULT IN LOSS OF POWER TO PILOT'S RHC, SBTC, RPTA, BFC, AND NWS. SHOULD THIS OCCUR, ONLY AUTOMATIC FLIGHT CONTROL MODE REMAINS AVAILABLE TO THE CREW.

- APPROVALS -

SS&PAE MANAGER

SS&PAE

DESIGN ENGINEERING

MEDS SYSTEM MEDS HARDWARE

JSC MOD

: P. STENGER-NGUYEN

: T. AL

: T. D. NGUYEN

: M. B. WARNER

. IVI. D. YVARINER

: R. SITAPARA

Ramone Stude 4-18-38